

SUPPLEMENTAL TYPE CERTIFICATE

10064806

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

ELECTRONICS INTERNATIONAL, Inc.

**63296 POWELL BUTTE HIGHWAY
BEND OR 97701
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: SEE FAA APPROVED MODEL LIST
(AML) dated 7.7. 2011 Rev. IR

Type Certificate Holder: SEE FAA APPROVED MODEL LIST

Type: SEE FAA APPROVED MODEL LIST

Model: SEE FAA APPROVED MODEL LIST

Original STC Number: FAA STC SA02135SE

Description of Design Change:

Installation of Electronics International Inc. Glass Panel Engine Monitor (with In-Flight Data) MVP-50T

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

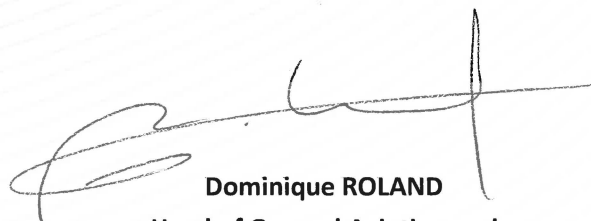
Associated Technical Documentation:

1. Installation Instructions, Document No. II 1211061, Revision D, dated 15 June 2011 or later FAA approved revision.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 01 March 2018



Dominique ROLAND

**Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)**



2. Airplane Flight Manual Supplement (AFMS) Document Number AFM06271101, Revision IR, dated 27 June 2011, or later FAA approved revisions.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List SA02135SE, dated 7 July 2011, Rev. IR or later approved revisions.

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